

## TORQUE AND TENSION IN FASTENERS

**Tightening Bolts:** Bolts are often tightened by applying torque to the head or nut, which causes the bolt to stretch. The stretching results in bolt tension or preload, which is the force that holds a joint together. Torque is relatively easy to measure with a torque wrench, so it is the most frequently used indicator of bolt tension. Unfortunately, a torque wrench does not measure bolt tension accurately, mainly because it does not take friction into account. The friction depends on bolt, nut, and washer material, surface smoothness, machining accuracy, degree of lubrication, and the number of times a bolt has been installed. Fastener manufacturers often provide information for determining torque requirements for tightening various bolts, accounting for friction and other effects. If this information is not available, the methods described in what follows give general guidelines for determining how much tension should be present in a bolt, and how much torque may need to be applied to arrive at that tension.

High preload tension helps keep bolts tight, increases joint strength, creates friction between parts to resist shear, and improves the fatigue resistance of bolted connections. The recommended preload  $F_i$ , which can be used for either static (stationary) or fatigue (alternating) applications, can be determined from:  $F_i = 0.75 \times A_t \times S_p$  for reusable connections, and  $F_i = 0.9 \times A_t \times S_p$  for permanent connections. In these formulas,  $F_i$  is the bolt preload,  $A_t$  is the tensile stress area of the bolt, and  $S_p$  is the proof strength of the bolt. Determine  $A_t$  from screw-thread tables or by means of formulas in this section. Proof strength  $S_p$  of commonly used ASTM and SAE steel fasteners is given in this section and in the section on metric screws and bolts for those fasteners. For other materials, an approximate value of proof strength can be obtained from:  $S_p = 0.85 \times S_y$ , where  $S_y$  is the yield strength of the material. Soft materials should not be used for threaded fasteners.

Once the required preload has been determined, one of the best ways to be sure that a bolt is properly tensioned is to measure its tension directly with a strain gage. Next best is to measure the change in length (elongation) of the bolt during tightening, using a micrometer or dial indicator. Each of the following two formulas calculates the required change in length of a bolt needed to make the bolt tension equal to the recommended preload. The change in length  $\delta$  of the bolt is given by:

$$\delta = F_i \times \frac{A_d \times l_t + A_t \times l_d}{A_d \times A_t \times E} \quad (1) \quad \text{or} \quad \delta = \frac{F_i \times l}{A \times E} \quad (2)$$

In Equation (1),  $F_i$  is the bolt preload;  $A_d$  is the major-diameter area of the bolt;  $A_t$  is the tensile-stress area of the bolt;  $E$  is the bolt modulus of elasticity;  $l_t$  is the length of the threaded portion of the fastener within the grip; and  $l_d$  is the length of the unthreaded portion of the grip. Here, the grip is defined as the total thickness of the clamped material. Equation (2) is a simplified formula for use when the area of the fastener is constant, and gives approximately the same results as Equation (1). In Equation (2),  $l$  is the bolt length;  $A$  is the bolt area; and  $\delta$ ,  $F_i$ , and  $E$  are as described before.

If measuring bolt elongation is not possible, the torque necessary to tighten the bolt must be estimated. If the recommended preload is known, use the following general relation for the torque:  $T = K \times F_i \times d$ , where  $T$  is the wrench torque,  $K$  is a constant that depends on the bolt material and size,  $F_i$  is the preload, and  $d$  is the nominal bolt diameter. A value of  $K = 0.2$  may be used in this equation for mild-steel bolts in the size range of  $\frac{1}{4}$  to 1 inch. For other steel bolts, use the following values of  $K$ : nonplated black finish, 0.3; zinc-plated, 0.2; lubricated, 0.18; cadmium-plated, 0.16. Check with bolt manufacturers and suppliers for values of  $K$  to use with bolts of other sizes and materials.

The proper torque to use for tightening bolts in sizes up to about  $\frac{1}{2}$  inch may also be determined by trial. Test a bolt by measuring the amount of torque required to fracture it (use bolt, nut, and washers equivalent to those chosen for the real application). Then, use a tightening torque of about 50 to 60 per cent of the fracture torque determined by the test. The tension in a bolt tightened using this procedure will be about 60 to 70 per cent of the elastic limit (yield strength) of the bolt material.

The table that follows can be used to get a rough idea of the torque necessary to properly tension a bolt by using the bolt diameter  $d$  and the coefficients  $b$  and  $m$  from the table; the approximate tightening torque  $T$  in ft-lb for the listed fasteners is obtained by solving the equation  $T = 10^{b+m \log d}$ . This equation is approximate, for use with unlubricated fasteners as supplied by the mill. See the notes at the end of the table for more details on using the equation.

**Wrench Torque  $T = 10^{b+m \log d}$  for Steel Bolts, Studs, and Cap Screws (see notes)**

Fastener Grade(s)	Bolt Diameter $d$ (in.)	$m$	$b$
SAE 2, ASTM A307	$\frac{1}{4}$ to 3	2.940	2.533
SAE 3	$\frac{1}{4}$ to 3	3.060	2.775
ASTM A-449, A-354-BB, SAE 5	$\frac{1}{4}$ to 3	2.965	2.759
ASTM A-325 <sup>a</sup>	$\frac{1}{2}$ to $1\frac{1}{2}$	2.922	2.893
ASTM A-354-BC	$\frac{1}{4}$ to $\frac{5}{8}$	3.046	2.837
SAE 6, SAE 7	$\frac{1}{4}$ to 3	3.095	2.948
SAE 8	$\frac{1}{4}$ to 3	3.095	2.983
ASTM A-354-BD, ASTM A490 <sup>a</sup>	$\frac{3}{8}$ to $1\frac{3}{4}$	3.092	3.057
Socket Head Cap Screws	$\frac{1}{4}$ to 3	3.096	3.014

<sup>a</sup> Values for permanent fastenings on steel structures.

Usage: Values calculated using the preceding equation are for standard, unplated industrial fasteners as received from the manufacturer; for cadmium-plated cap screws, multiply the torque by 0.9; for cadmium-plated nuts and bolts, multiply the torque by 0.8; for fasteners used with special lubricants, multiply the torque by 0.9; for studs, use cap screw values for equivalent grade.

**Preload for Bolts in Loaded Joints.**—The following recommendations are based on MIL-HDBK-60, a subsection of FED-STD-H28, Screw Thread Standards for Federal Service. Generally, bolt preload in joints should be high enough to maintain joint members in contact and in compression. Loss of compression in a joint may result in leakage of pressurized fluids past compression gaskets, loosening of fasteners under conditions of cyclic loading, and reduction of fastener fatigue life.

The relationship between fastener fatigue life and fastener preload is illustrated by Fig. 1. An axially loaded bolted joint in which there is no bolt preload is represented by line OAB, that is, the bolt load is equal to the joint load. When joint load varies between  $P_a$  and  $P_b$ , the bolt load varies accordingly between  $P_{Ba}$  and  $P_{Bb}$ . However, if preload  $P_{B1}$  is applied to the bolt, the joint is compressed and bolt load changes more slowly than the joint load (indicated by line  $P_{B1}A$ , whose slope is less than line OAB) because some of the load is absorbed as a reduction of compression in the joint. Thus, the axial load applied to the joint varies between  $P_{Ba'}$  and  $P_{Bb'}$  as joint load varies between  $P_a$  and  $P_b$ . This condition results in a considerable reduction in cyclic bolt-load variation and thereby increases the fatigue life of the fastener.

**Preload for Bolts In Shear.**—In shear-loaded joints, with members that slide, the joint members transmit shear loads to the fasteners in the joint and the preload must be sufficient to hold the joint members in contact. In joints that do not slide (i.e., there is no relative motion between joint members), shear loads are transmitted within the joint by frictional forces that mainly result from the preload. Therefore, preload must be great enough for the resulting friction forces to be greater than the applied shear force. With high applied shear loads, the shear stress induced in the fastener during application of the preload must also be

considered in the bolted-joint design. Joints with combined axial and shear loads must be analyzed to ensure that the bolts will not fail in either tension or shear.

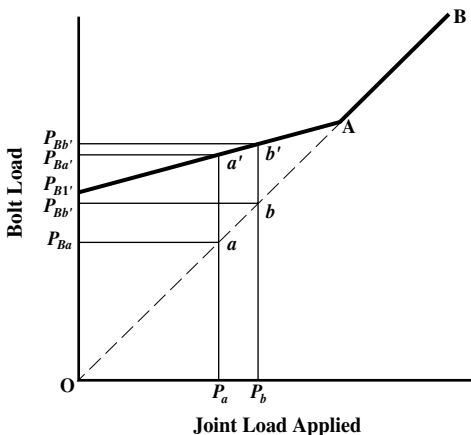


Fig. 1. Bolt Load in a Joint with Applied Axial Load

**General Application of Preload.**—Preload values should be based on joint requirements, as outlined before. Fastener applications are generally designed for maximum utilization of the fastener material; that is to say, the fastener size is the minimum required to perform its function and a maximum safe preload is generally applied to it. However, if a low-strength fastener is replaced by one of higher strength, for the sake of convenience or standardization, the preload in the replacement should not be increased beyond that required in the original fastener.

To utilize the maximum amount of bolt strength, bolts are sometimes tightened to or beyond the yield point of the material. This practice is generally limited to ductile materials, where there is considerable difference between the yield strength and the ultimate (breaking) strength, because low-ductility materials are more likely to fail due to unexpected overloads when preloaded to yield. Joints designed for primarily static load conditions that use ductile bolts, with a yield strain that is relatively far from the strain at fracture, are often preloaded above the yield point of the bolt material. Methods for tightening up to and beyond the yield point include tightening by feel without special tools, and the use of electronic equipment designed to compare the applied torque with the angular rotation of the fastener and detect changes that occur in the elastic properties of fasteners at yield.

Bolt loads are maintained below the yield point in joints subjected to cyclic loading and in joints using bolts of high-strength material where the yield strain is close to the strain at fracture. For these conditions, the maximum preloads generally fall within the following ranges: 50 to 80 per cent of the minimum tensile ultimate strength; 75 to 90 per cent of the minimum tensile yield strength or proof load; or 100 per cent of the observed proportional limit or onset of yield.

Bolt heads, driving recesses (in socket screws, for example), and the juncture of head and shank must be sufficiently strong to withstand the preload and any additional stress encountered during tightening. There must also be sufficient thread to prevent stripping (generally, at least three fully engaged threads). Materials susceptible to stress-corrosion cracking may require further preload limitations.

**Preload Adjustments.**—Preloads may be applied directly by axial loading or indirectly by turning of the nut or bolt. When preload is applied by turning of nuts or bolts, a torsion load component is added to the desired axial bolt load. This combined loading increases the tensile stress on the bolt. It is frequently assumed that the additional torsion load component dissipates quickly after the driving force is removed and, therefore, can be largely ignored. This assumption may be reasonable for fasteners loaded near to or beyond yield strength, but for critical applications where bolt tension must be maintained below yield, it is important to adjust the axial tension requirements to include the effects of the preload torsion. For this adjustment, the combined tensile stress (*von Mises* stress)  $F_{tc}$  in psi (MPa) can be calculated from the following:

$$F_{tc} = \sqrt{F_t^2 + 3F_s^2} \quad (3)$$

where  $F_t$  is the axial applied tensile stress in psi (MPa), and  $F_s$  is the shear stress in psi (MPa) caused by the torsion load application.

Some of the torsion load on a bolt, acquired when applying a preload, may be released by springback when the wrenching torque is removed. The amount of relaxation depends on the friction under the bolt head or nut. With controlled back turning of the nut, the torsional load may be reduced or eliminated without loss of axial load, reducing bolt stress and lowering creep and fatigue potential. However, calculation and control of the back-turn angle is difficult, so this method has limited application and cannot be used for short bolts because of the small angles involved.

For relatively soft work-hardenable materials, tightening bolts in a joint slightly beyond yield will work-harden the bolt to some degree. Back turning of the bolt to the desired tension will reduce embedment and metal flow and improve resistance to preload loss.

The following formula for use with single-start Unified inch screw threads calculates the combined tensile stress,  $F_{tc}$ :

$$F_{tc} = F_t \sqrt{1 + 3 \left( \frac{1.96 + 2.31\mu}{1 - 0.325P/d_2} - 1.96 \right)^2} \quad (4)$$

Single-start UNJ screw threads in accordance with MIL-S-8879 have a thread stress diameter equal to the bolt pitch diameter. For these threads,  $F_{tc}$  can be calculated from:

$$F_{tc} = F_t \sqrt{1 + 3 \left( \frac{0.637P}{d_2} + 2.31\mu \right)^2} \quad (5)$$

where  $\mu$  is the coefficient of friction between threads,  $P$  is the thread pitch ( $P = 1/n$ , and  $n$  is the number of threads per inch), and  $d_2$  is the bolt-thread pitch diameter in inches. Both Equations (2) and (3) are derived from Equation (1); thus, the quantity within the radical ( $\sqrt{\quad}$ ) represents the proportion of increase in axial bolt tension resulting from preload torsion. In these equations, tensile stress due to torsion load application becomes most significant when the thread friction,  $\mu$ , is high.

**Coefficients of Friction for Bolts and Nuts.**—Table 1 gives examples of coefficients of friction that are frequently used in determining torque requirements. Dry threads, indicated by the words "None added" in the Lubricant column, are assumed to have some residual machine oil lubrication. Table 1 values are not valid for threads that have been cleaned to remove all traces of lubrication because the coefficient of friction of these threads may be very much higher unless a plating or other film is acting as a lubricant.

**Table 1. Coefficients of Friction of Bolts and Nuts**

Bolt/Nut Materials	Lubricant	Coefficient of Friction, $\mu \pm 20\%$
Steel <sup>a</sup>	Graphite in petrolatum or oil	0.07
	Molybdenum disulfide grease	0.11
	Machine oil	0.15
Steel, <sup>a</sup> cadmium-plated	None added	0.12
Steel, <sup>a</sup> zinc-plated	None added	0.17
Steel <sup>a</sup> /bronze	None added	0.15
Corrosion-resistant steel or nickel-base alloys/silver-plated materials	None added	0.14
Titanium/steel <sup>a</sup>	Graphite in petrolatum	0.08
Titanium	Molybdenum disulfide grease	0.10

<sup>a</sup>“Steel” includes carbon and low-alloy steels but not corrosion-resistant steels.

Where two materials are separated by a slash (/), either may be the bolt material; the other is the nut material.

**Preload Relaxation.**—Local yielding, due to excess bearing stress under nuts and bolt heads (caused by high local spots, rough surface finish, and lack of perfect squareness of bolt and nut bearing surfaces), may result in preload relaxation after preloads are first applied to a bolt. Bolt tension also may be unevenly distributed over the threads in a joint, so thread deformation may occur, causing the load to be redistributed more evenly over the threaded length. Preload relaxation occurs over a period of minutes to hours after the application of the preload, so retightening after several minutes to several days may be required. As a general rule, an allowance for loss of preload of about 10 per cent may be made when designing a joint.

Increasing the resilience of a joint will make it more resistant to local yielding, that is, there will be less loss of preload due to yielding. When practical, a joint-length to bolt-diameter ratio of 4 or more is recommended (e.g., a ¼-inch bolt and a 1-inch or greater joint length). Through bolts, far-side tapped holes, spacers, and washers can be used in the joint design to improve the joint-length to bolt-diameter ratio.

Over an extended period of time, preload may be reduced or completely lost due to vibration; temperature cycling, including changes in ambient temperature; creep; joint load; and other factors. An increase in the initial bolt preload or the use of thread-locking methods that prevent relative motion of the joint may reduce the problem of preload relaxation due to vibration and temperature cycling. Creep is generally a high-temperature effect, although some loss of bolt tension can be expected even at normal temperatures. Harder materials and creep-resistant materials should be considered if creep is a problem or high-temperature service of the joint is expected.

The mechanical properties of fastener materials vary significantly with temperature, and allowance must be made for these changes when ambient temperatures range beyond 30 to 200°F. Mechanical properties that may change include tensile strength, yield strength, and modulus of elasticity. Where bolts and flange materials are generically dissimilar, such as carbon steel and corrosion-resistant steel or steel and brass, differences in thermal expansion that might cause preload to increase or decrease must be taken into consideration.

**Methods of Applying and Measuring Preload.**—Depending on the tightening method, the accuracy of preload application may vary up to 25 per cent or more. Care must be taken to maintain the calibration of torque and load indicators. Allowance should be made for uncertainties in bolt load to prevent overstressing the bolts or failing to obtain sufficient preload. The method of tensioning should be based on the required accuracy and relative costs.

The most common methods of bolt tension control are indirect because it is usually difficult or impractical to measure the tension produced in each fastener during assembly. Table 2 lists the most frequently used methods of applying bolt preload and the approximate accuracy of each method. For many applications, fastener tension can be satisfactorily controlled within certain limits by applying a known torque to the fastener. Laboratory tests have shown that whereas a satisfactory torque tension relationship can be established for a given set of conditions, a change of any of the variables, such as fastener material, surface finish, and the presence or absence of lubrication, may severely alter the relationship. Because most of the applied torque is absorbed in intermediate friction, a change in the surface roughness of the bearing surfaces or a change in the lubrication will drastically affect the friction and thus the torque tension relationship. Regardless of the method or accuracy of applying the preload, tension will decrease in time if the bolt, nut, or washer seating faces deform under load, if the bolt stretches or creeps under tensile load, or if cyclic loading causes relative motion between joint members.

**Table 2. Accuracy of Bolt Preload Application Methods**

Method	Accuracy	Method	Accuracy
By feel	±35%	Computer-controlled wrench	
Torque wrench	±25%	below yield (turn-of-nut)	±15%
Turn-of-nut	±15%	yield-point sensing	±8%
Preload indicating washer	±10%	Bolt elongation	±3–5%
Strain gages	±1%	Ultrasonic sensing	±1%

Tightening methods using power drivers are similar in accuracy to equivalent manual methods.

**Elongation Measurement.**—Bolt elongation is directly proportional to axial stress when the applied stress is within the elastic range of the material. If both ends of a bolt are accessible, a micrometer measurement of bolt length made before and after the application of tension will ensure the required axial stress is applied. The elongation  $\delta$  in inches (mm) can be determined from the formula  $\delta = F_t \times L_B \div E$ , given the required axial stress  $F_t$  in psi (MPa), the bolt modulus of elasticity  $E$  in psi (MPa), and the effective bolt length  $L_B$  in inches (mm).  $L_B$ , as indicated in Fig. 2, includes the contribution of bolt area and ends (head and nut) and is calculated from:

$$L_B = \left(\frac{d_{ts}}{d}\right)^2 \times \left(L_s + \frac{H_B}{2}\right) + L_J - L_s + \frac{H_N}{2} \quad (6)$$

where  $d_{ts}$  is the thread stress diameter,  $d$  is the bolt diameter,  $L_s$  is the unthreaded length of the bolt shank,  $L_J$  is the overall joint length,  $H_B$  is the height of the bolt head, and  $H_N$  is the height of the nut.

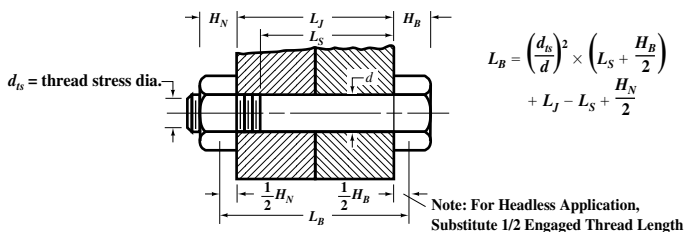


Fig. 2. Effective Length Applicable in Elongation Formulas

The micrometer method is most easily and accurately applied to bolts that are essentially uniform throughout the bolt length, that is, threaded along the entire length or that have only a few threads in the bolt grip area. If the bolt geometry is complex, such as tapered or stepped, the elongation is equal to the sum of the elongations of each section with allowances made for transitional stresses in bolt head height and nut engagement length.

The direct method of measuring elongation is practical only if both ends of a bolt are accessible. Otherwise, if the diameter of the bolt or stud is sufficiently large, an axial hole can be drilled, as shown in Fig. 3, and a micrometer depth gage or other means used to determine the change in length of the hole as the fastener is tightened. A similar method uses a special indicating bolt that has a blind axial hole containing a pin fixed at the bottom. The pin is usually made flush with the bolt head surface before load application. As the bolt is loaded, the elongation causes the end of the pin to move below the reference surface. The displacement of the pin can be converted directly into unit stress by means of a calibrated gage. In some bolts of this type, the pin is set a distance above the bolt so that the pin is flush with the bolt head when the required axial load is reached.

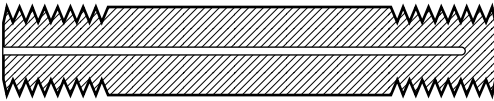


Fig. 3. Hole Drilled to Measure Elongation When One End of Stud or Bolt Is Not Accessible

The *ultrasonic method* of measuring elongation uses a sound pulse, generated at one end of a bolt, that travels the length of a bolt, bounces off the far end, and returns to the sound generator in a measured period of time. The time required for the sound pulse to return depends on the length of the bolt and the speed of sound in the bolt material. The speed of sound in the bolt depends on the material, the temperature, and the stress level. The ultrasonic measurement system can compute the stress, load, or elongation of the bolt at any time by comparing the pulse travel time in the loaded and unstressed conditions. In a similar method, measuring round-trip transit times of longitudinal and shear wave sonic pulses allows calculation of tensile stress in a bolt without consideration of bolt length. This method permits checking bolt tension at any time and does not require a record of the ultrasonic characteristics of each bolt at zero load.

To ensure consistent results, the ultrasonic method requires that both ends of the bolt be finished square to the bolt axis. The accuracy of ultrasonic measurement compares favorably with strain gage methods, but is limited by sonic velocity variations between bolts of the same material and by corrections that must be made for unstressed portions of the bolt heads and threads.

The *turn-of-nut method* applies preload by turning a nut through an angle that corresponds to a given elongation. The elongation of the bolt is related to the angle turned by the formula:  $\delta_B = \theta \times l \div 360$ , where  $\delta_B$  is the elongation in inches (mm),  $\theta$  is the turn angle of the nut in degrees, and  $l$  is the lead of the thread helix in inches (mm). Substituting  $F_t \times L_B \div E$  for elongation  $\delta_B$  in this equation gives the turn-of-nut angle required to attain preload  $F_t$ :

$$\theta = 360 \frac{F_t L_B}{E l} \quad (7)$$

where  $L_B$  is given by Equation (6), and  $E$  is the modulus of elasticity.

Accuracy of the turn-of-nut method is affected by elastic deformation of the threads, by roughness of the bearing surfaces, and by the difficulty of determining the starting point for measuring the angle. The starting point is usually found by tightening the nut enough to seat the contact surfaces firmly, and then loosening it just enough to release any tension and twisting in the bolt. The nut-turn angle will be different for each bolt size, length, mate-

rial, and thread lead. The preceding method of calculating the nut-turn angle also requires elongation of the bolt without a corresponding compression of the joint material. The turn-of-nut method, as just outlined, is not valid for joints with compressible gaskets or other soft material, or if there is a significant deformation of the nut and joint material relative to that of the bolt. The nut-turn angle would then have to be determined empirically using a simulated joint and a tension-measuring device.

The Japanese Industrial Standards (JIS) Handbook, *Fasteners and Screw Threads*, indicates that the turn-of-nut tightening method is applicable in both elastic and plastic region tightening. Refer to JIS B 1083 for more detail on this subject.

*Heating* causes a bolt to expand at a rate proportional to its coefficient of expansion. When a hot bolt and nut are fastened in a joint and cooled, the bolt shrinks and tension is developed. The temperature necessary to develop an axial stress,  $F_p$  (when the stress is below the elastic limit) can be found as follows:

$$T = \frac{F_t}{Ee} + T_o \quad (8)$$

In this equation,  $T$  is the temperature in degrees Fahrenheit needed to develop the axial tensile stress  $F_t$  in psi,  $E$  is the bolt material modulus of elasticity in psi,  $e$  is the coefficient of linear expansion in in./in.-°F, and  $T_o$  is the temperature in degrees Fahrenheit to which the bolt will be cooled.  $T - T_o$  is, therefore, the temperature change of the bolt. In finite-element simulations, heating and cooling are frequently used to preload mesh elements in tension or compression. Equation (8) can be used to determine required temperature changes in such problems.

*Example:* A tensile stress of 40,000 psi is required for a steel bolt in a joint operating at 70°F. If  $E$  is  $30 \times 10^6$  psi and  $e$  is  $6.2 \times 10^{-6}$  in./in.-°F, determine the temperature of the bolt needed to develop the required stress on cooling.

$$T = \frac{40,000}{(30 \times 10^6)(6.2 \times 10^{-6})} + 70 = 285^\circ\text{F}$$

In practice, the bolt is heated slightly above the required temperature (to allow for some cooling while the nut is screwed down) and the nut is tightened snugly. Tension develops as the bolt cools. In another method, the nut is tightened snugly on the bolt, and the bolt is heated in place. When the bolt has elongated sufficiently, as indicated by inserting a thickness gage between the nut and the bearing surface of the joint, the nut is tightened. The bolt develops the required tension as it cools; however, preload may be lost if the joint temperature increases appreciably while the bolt is being heated.

**Calculating Thread Tensile-Stress Area.**—The tensile-stress area for Unified threads is based on a diameter equivalent to the mean of the pitch and minor diameters. The pitch and the minor diameters for Unified screw threads can be found from the major (nominal) diameter,  $d$ , and the screw pitch,  $P = 1/n$ , where  $n$  is the number of threads per inch, by use of the following formulas: the pitch diameter  $d_p = d - 0.649519 \times P$ ; the minor diameter  $d_m = d - 1.299038 \times P$ . The tensile stress area,  $A_s$ , for Unified threads can then be found as follows:

$$A_s = \frac{\pi}{4} \left( \frac{d_m + d_p}{2} \right)^2 \quad (9)$$

UNJ threads in accordance with MIL-S-8879 have a tensile thread area that is usually considered to be at the basic bolt pitch diameter, so for these threads,  $A_s = \pi d_p^2/4$ . The tensile stress area for Unified screw threads is smaller than this area, so the required tightening torque for UNJ threaded bolts is greater than for an equally stressed Unified threaded bolt



in an equivalent joint. To convert tightening torque for a Unified fastener to the equivalent torque required with a UNJ fastener, use the following relationship:

$$\text{UNJ}_{\text{torque}} = \left( \frac{d \times n - 0.6495}{d \times n - 0.9743} \right)^2 \times \text{Unified}_{\text{torque}} \quad (10)$$

where  $d$  is the basic thread major diameter, and  $n$  is the number of threads per inch.

The tensile stress area for metric threads is based on a diameter equivalent to the mean of the pitch diameter and a diameter obtained by subtracting  $\frac{1}{6}$  the height of the fundamental thread triangle from the external-thread minor diameter. The Japanese Industrial Standard JIS B 1082 (see also ISO 898/1) defines the stress area of metric screw threads as follows:

$$A_s = \frac{\pi}{4} \left( \frac{d_2 + d_3}{2} \right)^2 \quad (11)$$

In Equation (11),  $A_s$  is the stress area of the metric screw thread in  $\text{mm}^2$ ;  $d_2$  is the pitch diameter of the external thread in mm, given by  $d_2 = d - 0.649515 \times P$ ; and  $d_3$  is defined by  $d_3 = d_1 - H/6$ . Here,  $d$  is the nominal bolt diameter;  $P$  is the thread pitch;  $d_1 = d - 1.082532 \times P$  is the minor diameter of the external thread in mm; and  $H = 0.866025 \times P$  is the height of the fundamental thread triangle. Substituting the formulas for  $d_2$  and  $d_3$  into Equation (11) results in  $A_s = 0.7854(d - 0.9382P)^2$ .

The stress area,  $A_s$ , of Unified threads in  $\text{mm}^2$  is given in JIS B 1082 as:

$$A_s = 0.7854 \left( d - \frac{0.9743}{n} \times 25.4 \right)^2 \quad (12)$$

**Relation between Torque and Clamping Force.**—The Japanese Industrial Standard JIS B 1803 defines fastener tightening torque  $T_f$  as the sum of the bearing surface torque  $T_w$  and the shank (threaded) portion torque  $T_s$ . The relationship between the applied tightening torque and bolt preload  $F_f$  is as follows:  $T_f = T_s + T_w = K \times F_f \times d$ . In the preceding,  $d$  is the nominal diameter of the screw thread, and  $K$  is the torque coefficient defined as follows:

$$K = \frac{1}{2d} \left( \frac{P}{\pi} + \mu_s d_2 \sec \alpha' + \mu_w D_w \right) \quad (13)$$

where  $P$  is the screw thread pitch;  $\mu_s$  is the coefficient of friction between threads;  $d_2$  is the pitch diameter of the thread;  $\mu_w$  is the coefficient of friction between bearing surfaces;  $D_w$  is the equivalent diameter of the friction torque bearing surfaces; and  $\alpha'$  is the flank angle at the ridge perpendicular section of the thread ridge, defined by  $\tan \alpha' = \tan \alpha \cos \beta$ , where  $\alpha$  is the thread half angle ( $30^\circ$ , for example), and  $\beta$  is the thread helix, or lead, angle.  $\beta$  can be found from  $\tan \beta = l \div 2\pi r$ , where  $l$  is the thread lead, and  $r$  is the thread radius (i.e., one-half the nominal diameter  $d$ ). When the bearing surface contact area is circular,  $D_w$  can be obtained as follows:

$$D_w = \frac{2}{3} \times \frac{D_o^3 - D_i^3}{D_o^2 - D_i^2} \quad (14)$$

where  $D_o$  and  $D_i$  are the outside and inside diameters, respectively, of the bearing surface contact area.

The torques attributable to the threaded portion of a fastener,  $T_s$ , and bearing surfaces of a joint,  $T_w$ , are as follows:

$$T_s = \frac{F_f}{2} \left( \frac{P}{\pi} + \mu_s d_2 \sec \alpha' \right) \quad (15)$$

$$T_w = \frac{F_f}{2} \mu_w D_w \quad (16)$$

where  $F_{fy}$ ,  $P$ ,  $\mu$ ,  $d_2$ ,  $\alpha'$ ,  $\mu_w$ , and  $D_w$  are as previously defined.

Tables 3 and 4 give values of torque coefficient  $K$  for coarse- and fine-pitch metric screw threads corresponding to various values of  $\mu_s$  and  $\mu_w$ . When a fastener material yields according to the shearing-strain energy theory, the torque corresponding to the yield clamping force (see Fig. 4) is  $T_{fy} = K \times F_{fy} \times d$ , where the yield clamping force  $F_{fy}$  is given by:

$$F_{fy} = \frac{\sigma_y A_s}{\sqrt{1 + 3 \left[ \frac{2}{d_A} \left( \frac{P}{\pi} + \mu_s d_2 \sec \alpha' \right) \right]^2}} \quad (17)$$

**Table 3. Torque Coefficients  $K$  for Metric Hexagon Head Bolt and Nut Coarse Screw Threads**

Between Threads, $\mu_s$	Coefficient of Friction Between Bearing Surfaces, $\mu_w$									
	0.08	0.10	0.12	0.15	0.20	0.25	0.30	0.35	0.40	0.45
0.08	0.117	0.130	0.143	0.163	0.195	0.228	0.261	0.293	0.326	0.359
0.10	0.127	0.140	0.153	0.173	0.206	0.239	0.271	0.304	0.337	0.369
0.12	0.138	0.151	0.164	0.184	0.216	0.249	0.282	0.314	0.347	0.380
0.15	0.153	0.167	0.180	0.199	0.232	0.265	0.297	0.330	0.363	0.396
0.20	0.180	0.193	0.206	0.226	0.258	0.291	0.324	0.356	0.389	0.422
0.25	0.206	0.219	0.232	0.252	0.284	0.317	0.350	0.383	0.415	0.448
0.30	0.232	0.245	0.258	0.278	0.311	0.343	0.376	0.409	0.442	0.474
0.35	0.258	0.271	0.284	0.304	0.337	0.370	0.402	0.435	0.468	0.500
0.40	0.285	0.298	0.311	0.330	0.363	0.396	0.428	0.461	0.494	0.527
0.45	0.311	0.324	0.337	0.357	0.389	0.422	0.455	0.487	0.520	0.553

Values in the table are average values of torque coefficient calculated using: Equations (13) and (14) for  $K$  and  $D_w$ ; diameters  $d$  of 4, 5, 6, 8, 10, 12, 16, 20, 24, 30, and 36 mm; and selected corresponding pitches  $P$  and pitch diameters  $d_2$  according to JIS B 0205 (ISO 724) thread standard. Dimension  $D_i$  was obtained for a Class 2 fit without chamfer from JIS B 1001, Diameters of Clearance Holes and Counterbores for Bolts and Screws (equivalent to ISO 273-1979). The value of  $D_o$  was obtained by multiplying the reference dimension from JIS B 1002, width across the flats of the hexagon head, by 0.95.

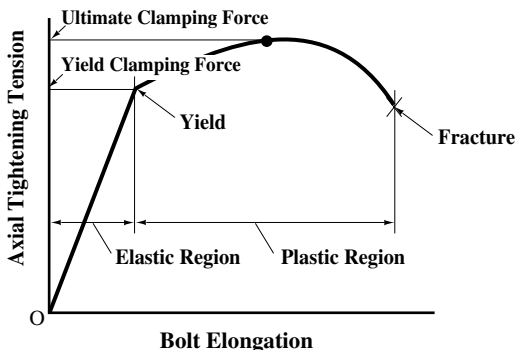


Fig. 4. The Relationship between Bolt Elongation and Axial Tightening Tension

**Table 4. Torque Coefficients  $K$  for Metric Hexagon Head Bolt and Nut Fine-Screw Threads**

Between Threads, $\mu_s$	Coefficient of Friction Between Bearing Surfaces, $\mu_w$									
	0.08	0.10	0.12	0.15	0.20	0.25	0.30	0.35	0.40	0.45
0.08	0.106	0.118	0.130	0.148	0.177	0.207	0.237	0.267	0.296	0.326
0.10	0.117	0.129	0.141	0.158	0.188	0.218	0.248	0.278	0.307	0.337
0.12	0.128	0.140	0.151	0.169	0.199	0.229	0.259	0.288	0.318	0.348
0.15	0.144	0.156	0.168	0.186	0.215	0.245	0.275	0.305	0.334	0.364
0.20	0.171	0.183	0.195	0.213	0.242	0.272	0.302	0.332	0.361	0.391
0.25	0.198	0.210	0.222	0.240	0.270	0.299	0.329	0.359	0.389	0.418
0.30	0.225	0.237	0.249	0.267	0.297	0.326	0.356	0.386	0.416	0.445
0.35	0.252	0.264	0.276	0.294	0.324	0.353	0.383	0.413	0.443	0.472
0.40	0.279	0.291	0.303	0.321	0.351	0.381	0.410	0.440	0.470	0.500
0.45	0.306	0.318	0.330	0.348	0.378	0.408	0.437	0.467	0.497	0.527

Values in the table are average values of torque coefficient calculated using Equations (13) and (14) for  $K$  and  $D_w$ ; diameters  $d$  of 8, 10, 12, 16, 20, 24, 30, and 36 mm; and selected respective pitches  $P$  and pitch diameters  $d_2$  according to JIS B 0207 thread standard (ISO 724). Dimension  $D_i$  was obtained for a Class 1 fit without chamfer from JIS B 1001, Diameters of Clearance Holes and Counterbores for Bolts and Screws (equivalent to ISO 273-1979). The value of  $D_o$  was obtained by multiplying the reference dimension from JIS B 1002 (small type series), width across the flats of the hexagon head, by 0.95.

In Equation (17),  $\sigma_y$  is the yield point or proof stress of the bolt,  $A_s$  is the stress area of the thread, and  $d_A = (4A_s/\pi)^{1/2}$  is the diameter of a circle having an area equal to the stress area of the thread. The other variables have been identified previously.

**Example:** Find the torque required to tighten a 10-mm coarse-threaded ( $P = 1.5$ ) grade 8.8 bolt to yield assuming that both the thread- and bearing-friction coefficients are 0.12.

**Solution:** From Equation (17), calculate  $F_{fy}$  and then solve  $T_{fy} = K F_{fy} d$  to obtain the torque required to stress the bolt to the yield point.

$$\sigma_y = 800 \text{ N/mm}^2 \text{ (MPa) (minimum, based on 8.8 grade rating)}$$

$$A_s = 0.7854(10 - 0.9382 \times 1.5)^2 = 57.99 \text{ mm}^2$$

$$d_A = (4A_s/\pi)^{1/2} = 8.6 \text{ mm}$$

$$d_2 = 9.026 \text{ mm (see JIS B 0205 or ISO 724)}$$

Find  $\alpha'$  from  $\tan \alpha' = \tan \alpha \cos \beta$  using:

$$\alpha = 30^\circ; \tan \beta = l \div 2\pi r; l = P = 1.5; \text{ and } r = d \div 2 = 5 \text{ mm}$$

$$\tan \beta = 1.5 \div 10\pi = 0.048, \text{ therefore } \beta = 2.73^\circ$$

$$\tan \alpha' = \tan \alpha \cos \beta = \tan 30^\circ \times \cos 2.73^\circ = 0.577, \text{ and } \alpha' = 29.97^\circ$$

Solving Equation (17) gives the yield clamping force as follows:

$$F_{fy} = \frac{800 \times 58.0}{\sqrt{1 + 3 \left[ \frac{2}{8.6} \left( \frac{1.5}{\pi} + 0.12 \times 9.026 \times \sec 29.97^\circ \right) \right]^2}} = 38,075 \text{ N}$$

$K$  can be determined from Tables 3 (coarse thread) and Tables 4 (fine thread) or from Equations (13) and (14). From Table 3, for  $\mu_s$  and  $\mu_w$  equal to 0.12,  $K = 0.164$ . The yield-point tightening torque can then be found from  $T_{fy} = K \times F_{fy} \times d = 0.164 \times 38,075 \times 10 = 62.4 \times 10^3 \text{ N-mm} = 62.4 \text{ N-m}$ .

**Obtaining Torque and Friction Coefficients.**—Given suitable test equipment, the torque coefficient  $K$  and friction coefficients between threads  $\mu_s$  or between bearing surfaces  $\mu_w$  can be determined experimentally as follows: Measure the value of the axial tight-

ening tension and the corresponding tightening torque at an arbitrary point in the 50 to 80 per cent range of the bolt yield point or proof stress (for steel bolts, use the minimum value of the yield point or proof stress multiplied by the stress area of the bolt). Repeat this test several times and average the results. The tightening torque may be considered as the sum of the torque on the threads plus the torque on the bolt head- or nut-to-joint bearing surface. The torque coefficient can be found from  $K = T_f \div F_f \times d$ , where  $F_f$  is the measured axial tension, and  $T_f$  is the measured tightening torque.

To measure the coefficient of friction between threads or bearing surfaces, obtain the total tightening torque and that portion of the torque due to the thread or bearing surface friction. If only tightening torque and the torque on the bearing surfaces can be measured, then the difference between these two measurements can be taken as the thread-tightening torque. Likewise, if only the tightening torque and threaded-portion torque are known, the torque due to bearing can be taken as the difference between the known torques. The coefficients of friction between threads and bearing surfaces, respectively, can be obtained from the following:

$$\mu_s = \frac{2T_s \cos \alpha'}{d_2 F_f} - \cos \alpha' \tan \beta \quad (18) \quad \mu_w = \frac{2T_w}{D_w F_f} \quad (19)$$

As before,  $T_s$  is the torque attributable to the threaded portion of the screw,  $T_w$  is the torque due to bearing,  $D_w$  is the equivalent diameter of friction torque on bearing surfaces according to Equation (14), and  $F_f$  is the measured axial tension.

**Torque-Tension Relationships.**—Torque is usually applied to develop an axial load in a bolt. To achieve the desired axial load in a bolt, the torque must overcome friction in the threads and friction under the nut or bolt head. In Fig. 5, the axial load  $P_B$  is a component of the normal force developed between threads. The normal-force component perpendicular to the thread helix is  $P_{N\beta}$  and the other component of this force is the torque load  $P_B \tan \beta$  that is applied in tightening the fastener. Assuming the turning force is applied at the pitch diameter of the thread, the torque  $T_1$  needed to develop the axial load is  $T_1 = P_B \times \tan \beta \times d_2/2$ . Substituting  $\tan \beta = l \div \pi d_2$  into the previous expression gives  $T_1 = P_B \times l \div 2\pi$ .

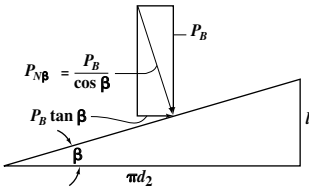


Fig. 5. Free Body Diagram of Thread Helix Forces

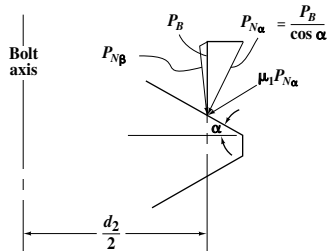


Fig. 6. Thread Friction Force

In Fig. 6, the normal-force component perpendicular to the thread flanks is  $P_{N\alpha}$ . With a coefficient of friction  $\mu_1$  between the threads, the friction load is equal to  $\mu_1 P_{N\alpha}$ , or  $\mu_1 P_B \div \cos \alpha$ . Assuming the force is applied at the pitch diameter of the thread, the torque  $T_2$  to overcome thread friction is given by:

$$T_2 = \frac{d_2 \mu_1 P_B}{2 \cos \alpha} \quad (20)$$

With the coefficient of friction  $\mu_2$  between a nut or bolt-head pressure face and a component face, as in Fig. 7, the friction load is equal to  $\mu_2 P_B$ . Assuming the force is applied midway between the nominal (bolt) diameter  $d$  and the pressure-face diameter  $b$ , the torque  $T_3$  to overcome the nut or bolt underhead friction is:

$$T_3 = \frac{d+b}{4} \mu_2 P_B \quad (21)$$

The total torque,  $T$ , required to develop axial bolt load,  $P_B$ , is equal to the sum of the torques  $T_1$ ,  $T_2$ , and  $T_3$  as follows:

$$T = P_B \left( \frac{l}{2\pi} + \frac{d_2 \mu_1}{2 \cos \alpha} + \frac{(d+b) \mu_2}{4} \right) \quad (22)$$

For a fastener system with  $60^\circ$  threads,  $\alpha = 30^\circ$  and  $d_2$  is approximately  $0.92d$ . If no loose washer is used under the rotated nut or bolt head,  $b$  is approximately  $1.5d$  and Equation (22) reduces to:

$$T = P_B [0.159 \times l + d(0.531 \mu_1 + 0.625 \mu_2)] \quad (23)$$

In addition to the conditions of Equation (23), if the thread and bearing friction coefficients,  $\mu_1$  and  $\mu_2$ , are equal (which is not necessarily so), then  $\mu_1 = \mu_2 = \mu$ , and the previous equation reduces to:

$$T = P_B (0.159l + 1.156 \mu d) \quad (24)$$

**Example:** Estimate the torque required to tighten a UNC  $\frac{1}{2}$ -13 grade 8 steel bolt to a preload equivalent to 55 per cent of the minimum tensile bolt strength. Assume that the bolt is unplated and both the thread and bearing friction coefficients equal 0.15.

**Solution:** The minimum tensile strength for SAE grade 8 bolt material is 150,000 psi (from page 1508). To use Equation (24), find the stress area of the bolt using Equation (9) with  $P = 1/13$ ,  $d_m = d - 1.2990P$ , and  $d_p = d - 0.6495P$ , and then calculate the necessary preload,  $P_B$ , and the applied torque,  $T$ .

$$A_s = \frac{\pi}{4} \left( \frac{0.4500 + 0.4001}{2} \right)^2 = 0.1419 \text{ in.}^2$$

$$P_B = \sigma_{\text{allow}} \times A_s = 0.55 \times 150,000 \times 0.1419 = 11,707 \text{ lb}_f$$

$$T = 11,707 \left( \frac{0.159}{13} + 1.156 \times 0.15 \times 0.500 \right) = 1158 \text{ lb-in.} = 96.5 \text{ lb-ft}$$

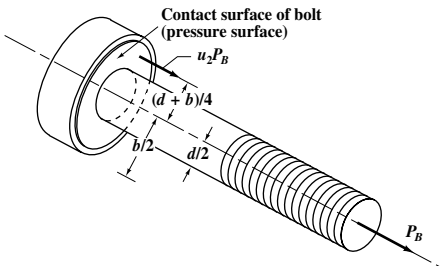


Fig. 7. Nut or Bolt Head Friction Force

**Grade Marks and Material Properties for Bolts and Screws.**—Bolts, screws, and other fasteners are marked on the head with a symbol that identifies the grade of the fastener. The grade specification establishes the minimum mechanical properties that the fastener must meet. Additionally, industrial fasteners must be stamped with a registered head mark that identifies the manufacturer. The grade identification table identifies the grade markings and gives mechanical properties for some commonly used ASTM and SAE steel fasteners. Metric fasteners are identified by property grade marks, which are specified in ISO and SAE standards. These marks are discussed with metric fasteners.

### Grade Identification Marks and Mechanical Properties of Bolts and Screws

Identifier	Grade	Size (in.)	Min. Strength (10 <sup>3</sup> psi)			Material & Treatment
			Proof	Tensile	Yield	
A	SAE Grade 1	$\frac{1}{4}$ to $1\frac{1}{2}$	33	60	36	1
	ASTM A307	$\frac{1}{4}$ to $1\frac{1}{2}$	33	60	36	3
	SAE Grade 2	$\frac{1}{4}$ to $\frac{3}{4}$	55	74	57	1
		$\frac{7}{8}$ to $1\frac{1}{2}$	33	60	36	
	SAE Grade 4	$\frac{1}{4}$ to $1\frac{1}{2}$	65	115	100	2, a
B	SAE Grade 5	$\frac{1}{4}$ to 1	85	120	92	2, b
	ASTM A449	$1\frac{1}{8}$ to $1\frac{1}{2}$	74	105	81	
	ASTM A449	$1\frac{3}{4}$ to 3	55	90	58	
C	SAE Grade 5.2	$\frac{1}{4}$ to 1	85	120	92	4, b
D	ASTM A325, Type 1	$\frac{1}{2}$ to 1	85	120	92	2, b
		$1\frac{1}{8}$ to $1\frac{1}{2}$	74	105	81	
E	ASTM A325, Type 2	$\frac{1}{2}$ to 1	85	120	92	4, b
		$1\frac{1}{8}$ to $1\frac{1}{2}$	74	105	81	
F	ASTM A325, Type 3	$\frac{1}{2}$ to 1	85	120	92	5, b
		$1\frac{1}{8}$ to $1\frac{1}{2}$	74	105	81	
G	ASTM A354, Grade BC	$\frac{1}{4}$ to $2\frac{1}{2}$	105	125	109	5, b
		$2\frac{3}{4}$ to 4	95	115	99	
H	SAE Grade 7	$\frac{1}{4}$ to $1\frac{1}{2}$	105	133	115	7, b
I	SAE Grade 8	$\frac{1}{4}$ to $1\frac{1}{2}$	120	150	130	7, b
	ASTM A354, Grade BD	$\frac{1}{4}$ to $1\frac{1}{2}$	120	150	130	6, b
J	SAE Grade 8.2	$\frac{1}{4}$ to 1	120	150	130	4, b
K	ASTM A490, Type 1	$\frac{1}{2}$ to $1\frac{1}{2}$	120	150	130	6, b
L	ASTM A490, Type 3					5, b

Material Steel: 1—low or medium carbon; 2—medium carbon; 3—low carbon; 4—low-carbon martensite; 5—weathering steel; 6—alloy steel; 7—medium-carbon alloy. Treatment: a—cold drawn; b—quench and temper.